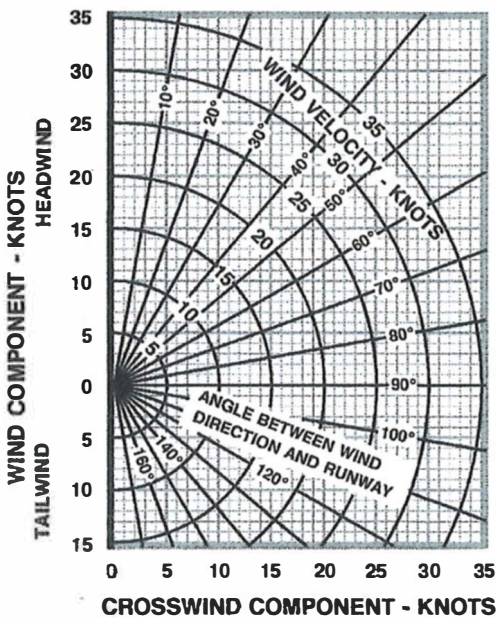


Name: _____

| | A | B | C | D |
|----|-----------------------------------|--------------------|---------------------|-----------------------------|
| 1 | | WEIGHT | ARM | MOMENT |
| 2 | EMPTY WEIGHT | | | |
| 3 | PILOT/FRONT PASSENGER | | 27.56 | |
| 4 | REAR PASSENGER | | 27.56 | |
| 5 | BAGGAGE AREA 1 | | 51.58 | |
| 6 | BAGGAGE AREA 2 Wing Locker | | 23.62 | |
| 7 | ZERO FUEL WIEGHT | | | |
| 8 | FUEL IN POUNDS | | 7.09 | |
| 9 | RAMP WEIGHT | | | |
| 10 | TAXI/RUN UP | | | |
| 11 | TAKEOFF WEIGHT | | | |
| 12 | FUEL BURN | | | |
| 13 | LANDING WEIGHT | | | |
| 14 | | | | |
| 15 | TAKEOFF CG : | | PRESS. ALT | |
| 16 | LANDING CG : | | DENSITY ALT. | |
| 17 | | | | |
| 18 | | GROUND ROLL | | OVER 50 FT. OBSTACLE |
| 19 | TAKE OFF DISTANCE : | | | |
| 20 | LANDING DISTANCE : | | | |

WIND COMPONENTS

NOTE:
Maximum demonstrated crosswind velocity is 16 knots (not a limitation).



$$\frac{\text{total moment}}{\text{total weight}} \times \frac{100}{59.06 (\text{MAC})} = \% \text{ of MAC}$$

